

The Use of Solid Ceramic and Ceramic Hard-Coated Components to Prolong the Performance of Perfluoropolyalkylether Lubricants

D. J. CARRÉ
Chemistry and Physics Laboratory
Laboratory Operations
The Aerospace Corporation
El Segundo, CA 90245-4691

30 September 1990

Prepared for

SPACE SYSTEMS DIVISION AIR FORCE SYSTEMS COMMAND Los Angeles Air Force Base P.O. Box 92960 Los Angeles, CA 90009-2960

APPROVED FOR PUBLIC RELEASE; DISTRIBUTION UNLIMITED



Approved for public releases
Distribution Unlimited

This report was submitted by The Aerospace Corporation, El Segundo, CA 90245, under Contract No. F04701-88-C-0089 with the Space Systems Division, P.O. Box 92960, Los Angeles, CA 90009-2960. It was reviewed and approved for The Aerospace Corporation by J. M. Straus, Director, Chemistry and Physics Laboratory. Lt Borden was the project officer for the Mission-Oriented Investigation and Experimentation (MOIE) Program.

This report has been reviewed by the Public Affairs Office (PAS) and is releasable to the National Technical Information Service (NTIS). At NTIS, it will be available to the general public, including foreign nationals.

This technical report has been reviewed and is approved for publication. Publication of this report does not constitute Air Force approval of the report's findings or conclusions. It is published only for the exchange and stimulation of ideas.

MARK BORDEN, 1ST Lt, USAF

MOIE Program Officer

SSD/MWBA

JONATHAN M. EMMES, Maj, USAI

MOIE Program Manager

AFSTC/WCO OL-AB

SECURITY CLASSIFICA	TION OF	THIS PAGE
---------------------	---------	-----------

REPORT DOCUMENTATION PAGE					
1a. REPORT SECURITY CLASSIFICATION	1b. RESTRICTIVE MARKINGS				
Inclassified					
2a. SECURITY CLASSIFICATION AUTHORITY		3. DISTRIBUTION/AVA	ILABILITY OF REP	ORT	
2b. DECLASSIFICATION/DOWNGRADING SCHEDULE		Approved for	public relea	ise;	
		distribution unlimited.			
4. PERFORMING ORGANIZATION REPORT NUMBER(S)		5. MONITORING ORG	ANIZATION REPO	RT NUMBER	R(S)
TR-0090(5945-03)-2 SSD-TR-90-32				[
6a. NAME OF PERFORMING ORGANIZATION 66	OFFICE SYMBOL				
The Aerospace Corporation	(If applicable)	Space Systems Division			Ì
Laboratory Operations					
6c. ADDRESS (City, State, and ZIP Code)		7b. ADDRESS (City, S Los Angeles A	ir Force Ba:	se	
El Segundo, CA 90245-4691		Los Angeles,	CA 90009-29	60	
8a. NAME OF FUNDING/SPONSORING 8	b. OFFICE SYMBOL	9. PROCUREMENT IN	ISTRUMENT IDEN	TIFICATION	NUMBER
ORGANIZATION	(If applicable)	F04701-88-C-0	089		
		and and their of FUN	5.00 to 10.0550		
8c. ADDRESS (City, State, and ZIP Code)		10. SOURCE OF FUN PROGRAM		ASK	WORK UNIT
		ELEMENT NO.		10.	ACCESSION NO.
Perfluoropolyalkylether Lubricants 12. PERSONAL AUTHOR(S) Carré, David J. 13a. TYPE OF REPORT 13b. TIME COVERED 14. DATE OF REPORT (Year, Month, Day) 15. PAGE COUNT 1990 September 30					
FROM	то		September 30		20
16. SUPPLEMENTARY NOTATION-					
17. COSATI CODES	18) SUBJECT TERMS	Continue on reverse it i Perfluoropol	necessary and iden	tity by block	number) ear testing
FIELD GROUP SUB-GROUP	Hard coatings				· · · · · · · · · · · · · · · · · · ·
	Lubricants	Titanium nit	ride		
19 ABSTRACT (Continue on reverse if necessary and in	lentify by block number				
19. ABSTRACT (Continue on reverse if necessary and identify by block number) The performance of perfluoropolyalkylether (PFPE) oils under boundary lubrication conditions was tested using 440C stainless steel, silicon nitride/440C hybrid, and titanium nitride					
hard-coated ball-bearing components. By limiting the chemical interactions between the lubricant and iron, the performance was extended by a factor of 5 to 10 using the ceramic materials instead of 440C components. The maximum performance enhancement for the PFPE oils under our test conditions was limited by thermal degradation at the high temperatures generated in the asperity contact region.					
20. DISTRIBUTION/AVAILABILITY OF ABSTRACT 21. ABSTRACT SECURITY CLASSIFICATION					
UNCLASSIFIED/UNLIMITED SAME AS RPT. DTIC USERS Unclassified					
22a. NAME OF RESPONSIBLE INDIVIDUAL 22b. TELEPHONE (Include Area Code) 22c. OFFICE SYMBOL			ICE SYMBOL		
1		1			

PREFACE

The author would like to thank Dr.~W.~D. Sproul for the TiN coatings and Dr.~M. Hilton for the scanning electron microscopy.



Acce	ssion For	· · · · · · · · · · · · · · · · · · ·		
NTIS	GRA&I	D		
DTIC	TAB	Ü		
Unan	nounced	H		
Just	ification_			
By				
Dist	Distribution/			
Avai	Availability Codes			
	Avail and	10		
Dist	Special			
1	1	•		
11	1 1	- 1		
r	1	1		
-	L	ı		

CONTENTS

PREFA	CE	1
Ι.	INTRODUCTION	5
11.	EXPERIMENTAL	9
	A. Materials B. Wear-Test Facility C. Experimental Procedure	9 9 12
III.	RESULTS AND DISCUSSION	15
IV.	CONCLUSIONS	21
REFER	ENCES	23
	FIGURES	
1.	Hypothesized Mechanism	Ė
2.	Wear-Test Fixture	11
3.	Wear Life as a Function of Axial Load	17
4.	Wear Life as a Function of Hertzian Stress	¹ ô
	TABLES	
1.	PFPE Oil Properties	10
2.	Wear Test Conditions	13
3	Wear Test Results	16

I. INTRODUCTION

Perfluoropolyalkylether (PFPE) synthetic oils and oil-based greases are relatively new lubricants. They are especially useful in spacecraft because of their low vapor pressure, high viscosity index, and general chemical inertness. However, the usefulness of PFPE lubricants is limited by their poor performance under boundary lubrication conditions [1,2].

In our laboratory, we have studied the degradation of PFPE oils under boundary conditions. We have demonstrated that, in the spacecraft orbital environment, i.e., low partial pressures of oxygen, the oils degrade at high temperatures by reaction with iron (III) fluoride, a Lewis acid catalyst [3]. The degradation reaction is measurable at 350°C, which is 30°C below the temperature where thermal degradation becomes important. Temperatures $\ge 350^{\circ}$ C are estimated to be generated at asperity contacts [4]. The degradation results in the production of acid fluoride chaincleavage products and a reduction in the average molecular weight of the oil. Furthermore, using surface spectroscopic techniques, we have observed iron fluoride compounds on the wear surfaces after bearing testing [5]. These results support the hypothesis that the PFPE oils interact with the iron in ball-bearing steels at asperity contact temperatures to produce iron (III) fluoride, which further degrades the PFPE through an autocatalytic pathway [1-3,5]. (The scheme for PFPE degradation is given in Fig. 1.) To extend the boundary lubrication performance and usefulness of the PFPE lubricants, the interactions between PFPE and iron metal at the lubricant/steel interface need to be reduced or eliminated.

One method for prolonging wear life under boundary conditions is to use antiwear additives in the oil. Finding appropriate additives for PFPE oils is difficult because of solubility constraints. However, work is currently under way to synthesize additives for PFPE lubricants [6]. An alternative method for prolonging wear life is to use nonferrous materials at the contacting interface, such as ceramic hard-coated or solid ceramic

STEP 2:

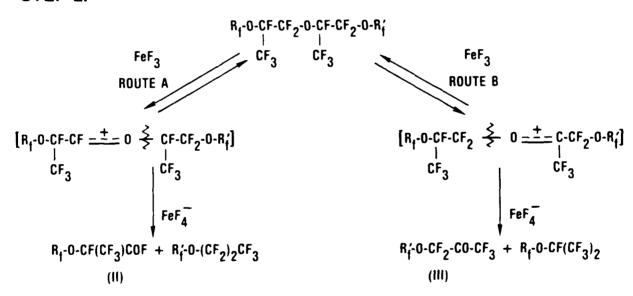


Fig. 1. Hypothesized Mechanism.

(Reprinted by permission of the Society of Tribologists and Lubrication Engineers. All rights reserved.)

bearing components. Ceramic hard-coated bearing components have been demonstrated to prolong the lifetime of instrument bearings under elastohydrodynamic (EHD) conditions [7-10]. Solid ceramic bearing components, such as ball bearings with titanium-carbide-coated tungstencarbide talls and zirconium-nitride-coated raceways, have extended the performance of a linear PFPE oil [11]. Using these nonferrous materials would circumvent the welding interactions in stainless steel components that expose chemically reactive iron metal during the wear process. In this report, we present the results of a series of wear tests using stainless steel, hybrid silicon nitride/stainless steel, and titanium nitride hard-coated stainless steel components with the PFPE oil.

II. EXPERIMENTAL

A. MATERIALS

The PFPE fluid, Krytox 143AB, was obtained from E. I. Du Pont Company and was used without further purification. This fluid is a branched compound that can be described by the following generalized formula:

$$R_f - (CF - CF_2O)_n - R_f'$$

$$CF_3$$

where $R_{\hat{f}}$ and $R_{\hat{f}}'$ are PFPE end groups of unspecified length and n is approximately 20. The properties of the fluid are given in Table 1.

Aetna F1 thrust ball bearings were used, in which the 4.72~mm (3/16~in.) balls were replaced by grade 10, 440C stainless steel balls (Winsted Precision Ball Company); grade 10, hot-pressed silicon nitride balls (Norton Company); or grade 10, 440C balls that had been reactively sputtered with a $0.75~\mu m$ TiN hard coating (Basic Industry Research Laboratory, Northwestern University). The thrust bearing raceways were used as received or were hard coated with TiN. The wear disks were 32 mm diam, 440C stainless steel, polished to a mean surface finish of $0.3~\mu m$. Wear disks with $0.75~\mu m$ TiN hard coating were also used.

B. WEAR-TEST FACILITY

The wear-test facility is shown in Fig. 2 and has been described in detail previously [5]. Boundary lubrication conditions were "forced" by the eccentric interaction between a flat disk, and the balls and lower raceway of a thrust bearing. The eccentricity was produced by noncoaxial alignment of the interacting members: the lower thrust-bearing raceway was positioned ~1 mm off axis. This resulted in a calculated skid/roll ratio of ~0.16. The tests were run at 1750 rpm, which corresponds to a relative linear velocity of 140 m/min. The tests began at ambient temperature; no

Table 1. PFPE Oil Properties

Viscosity	
37.8°C	$85 \text{ mm}^2 - \text{s}^{-1}$
98.9°C	$10.3 \text{ mm}^2 - \text{s}^{-1}$
Viscosity index	113
Pour point	-43°C
Density at 23.9°C	$1.89 \times 10^3 \text{ kg-m}^{-3}$
Surface tension at 26°C	$1.85 \times 10^{-2} \text{ N-m}^{-1}$
Vapor Pressure	
149°C	40.1 Pa
204°C	334 Pa
260 ^o c	1377 Pa
315°C	7019 Pa

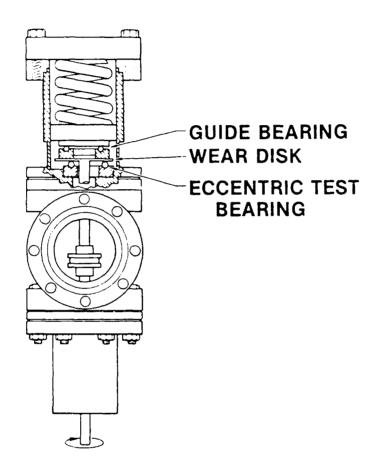


Fig. 2. Wear-Test Fixture.
(Reprinted by permission of the Society of Tribologists and Lubrication Eng.neers. All rights reserved.)

attempt was made to control the temperature. The test conditions are given in Table 2.

The disk/thrust bearing interaction members were housed in a stainless steel vacuum chamber that was pumped with a turbomolecular pump, so that pressures of $\leq 1.3 \times 10^{-3}$ Pa were routinely achieved, simulating the space environment.

C. EXPERIMENTAL PROCEDURE

The wear members were lubricated by applying 20 µl of a 50% (v/v) solution of the PFPE oil in Freon TF (1,1,2-trichloro-1,2,2-trifluoroethane) to the balls and allowing the solvent to evaporate before placing the wear disk in position. The status of the tests was determined by monitoring the torque indirectly by using the current draw of the motor, measured as the potential drop across a precision 1 Ω resistor in series with the motor. The torque of the wear members is the difference between the system torque and the torque in the absence of the wear members, which was independently determined. (The torque of the guide bearing was considered to be insignificant.) The tests were terminated when the torque increased to 150% of the initial steady-state value. (This value was arbitrarily chosen to represent system failure.)

Three different materials combinations were used to vary the relative rate of iron metal exposure resulting from wear processes: 440C balls/440C disk; Si_3N_4 balls/440C disk; and balls, disk, and raceway hard coated with TiN. These materials combinations would give iron exposure through "welding" interactions, less iron exposure because the welding interaction would be absent, and little or no iron exposure, respectively. (With the exception of the hard-coated combination, the original equipment, a 52100 stainless steel thrust-bearing lower raceway, was used. The Hertzian stress at the ball/raceway interface was much less than that at the ball/disk interface and was considered to be insignificant with respect to lubricant degradation.)

Table 2. Wear Test Conditions

 	
Speed	1750 rpm
Axial load	89-200 N
Pressure	<1.3 x 10 ^{−3} Pa
Temperature	ambient ^à
Lubricant quantity	10 µl

 $^{^{\}mathbf{a}}$ The temperature was not controlled.

Hertzian stress values were calculated using normal methods. The average stress over the contact area is used in this report [12]. For the TiN hard-coated components, the stress calculation should take into account both the coating and the substrate. However, van der Zwaag and Field [13] have shown that, as the ratio of circular contact radius to coating thickness increases, the calculated Hertzian stress approaches the value calculated from the substrate parameters only. Using Young's modulus values of 640 GPa for TiN [14] and 193 GPa for 440C stainless steel [12], and a radius:thickness ratio of 40, the Hertzian stress for the hard-coated system is within 5% of that calculated for the substrate alone [13]. Because the radius:thickness ratio was >40 for our system, only the substrate parameters were used to calculate the Hertzian stress, with the assumption that the stress of the hard-coated combination was within 5% of the calculated value and would not significantly affect the interpretation of the test results.

III. RESULTS AND DISCUSSION

The results of the wear tests are given in Table 3. Figure 3 is a plot of wear life as a function of applied axial load. As with wear tests in general, the run-to-run variation in wear life is rather large, which can be seen in the data in Table 3. The values shown are the average of at least three separate runs. In the plotted data, only the average values are shown, with a linear regression fit for each data set shown as a straight line. In spite of the large spread in the wear test values, the trends in the data are clear. The values for $Si_3N_4/440C$ have essentially the same wear life/load relationship as the values for 440C/440C, whereas the values for TiN/TiN are clearly different. For s lar applied loads, the TiN/TiN combination results in longer wear lives by a factor of 5 to 10. When plotted as in Fig. 3, the data for the $Si_3N_4/440C$ combination are presented in "biased" fashion, because the Hertzian stress for this combination is considerably higher than the stress for the 440C/440C combination at the same applied load. For engineering considerations, with fixed loads, it makes more sense to use the TiN hard coating than the $Si_{2}N_{11}/440C$ hybrid system. However, to understand the chemical processes that occur in the contact region, the Hertzian stress is a more meaningful parameter.

In Fig. 4, the wear-life data for the three materials combinations are plotted as a function of Hertzian stress. The TiN/TiN and 440C/440C combinations retain the same relative positions as in Fig. 3, because the Hertzian stress is calculated based on the parameters for the 440C substrate. However, the $\rm Si_3N_4/440C$ data are now grouped with the TiN/TiN data, with wear-life values 5-10 times larger than for the 440C/440C combination.

Our initial expectation was that the average wear-life values for the TiN/TiN combination would be significantly larger than those for the ${\rm Si}_3{\rm N}_4/440C$ combination. In the latter case, some metallic iron would be

Table 3. Wear Test Results

Applied Load (N)	Hertzian Stress (x10 ⁹ Pa)	Average Total Revolutions (x10 ⁶)	Standard Deviation (x10 ⁶)
440C/440C			
134	1.13	2.86	1.47
156	1.21	1.31	1.07
200	1.32	0.397	0.601
Si ₃ N ₄ /440C			
89	1.14	8.19	3.74
111	1.23	5.86	2.34
134	1.30	2.60	0.60
TiN/TiN			
134	1.13	5.66	3.47
156	1.21	8.91	5.53
200	1.32	3.44	1,15

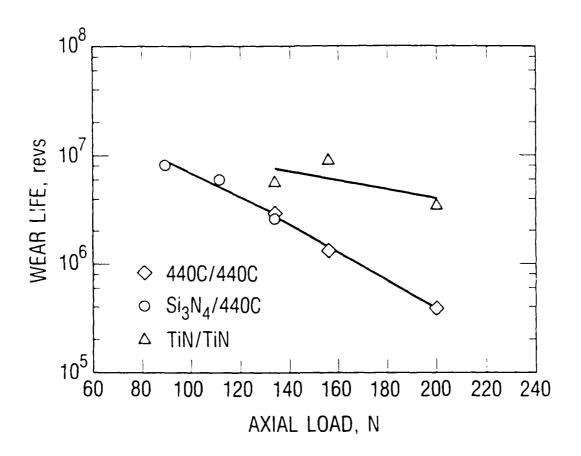


Fig. 3. Wear Life as a Function of Axial Load

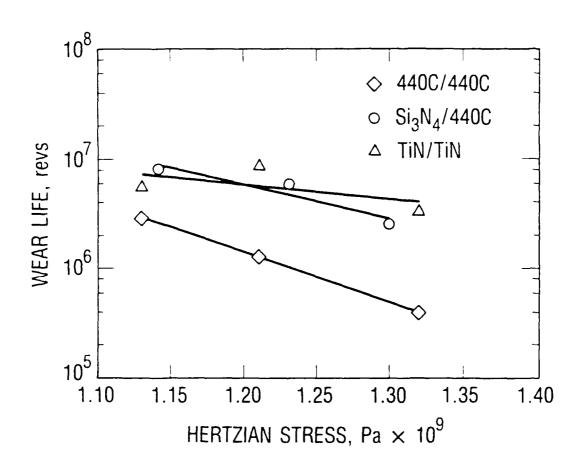


Fig. 4. Wear Life as a Function of Hertzian Stress

expected to be exposed during the wear process (some catalysis), whereas in the former case, no iron metal would be expected (no catalysis). The assumption that the Hertzian stress values for the TiN/TiN system are approximated using substrate parameters would not make a significant difference in the relative wear lives if the approximated values were within 5% of the actual values. We believe that the lack of wear-life enhancement with the TiN/TiN combination over the ${\rm Si}_3{\rm N}_4/440{\rm C}$ combination is the result of a kinetic effect.

In previous work, we had demonstrated that the FeF_3 catalyzed degradation of the PFPE fluid became kinetically significant at a temperature of 350°C and that the rate of thermal degradation was not comparable until a temperature of 380° C was reached [3]. If we assume that the thermal degradation rate doubles for each 10°C rise in temperature, then at 350°C, the thermal degradation rate would be slower than the Lewis acid-catalyzed reaction by approximately a factor of 8. (This assumption is used because activation energies, and thus, temperature dependencies, for the catalyzed and thermal degradation reactions are not available.) We conclude that by retarding the acid-catalyzed rate through the elimination or significant reduction in the amount of iron metal formed at the contact, the limiting reaction becomes thermal degradation. To the first order, thermal degradation is not influenced by the choice of materials. (In our investigations, we analyzed the surfaces of the TiN-coated components using scanning electron microscopy (data not shown) and could find no evidence for breakdown of the coating with concomitant exposure of iron.] The implication of this conclusion is that the performance of the PFPE fluids could only be improved by approximately a factor of 8 greater lifetime than that achievable with the 440C/440C system. This predicted enhancement is consistent with the 5-10 lifetime increase measured in our testing.

In addition to the hypothesized mechanism involving Lewis acidcatalyzed PFPE degradation, a mechanism involving reaction of the oxygen in the molecule with iron has also been proposed [11]. Because the interaction with exposed iron is common to both mechanisms, the reduction of the rate of iron exposure is consistent with longer wear lives in hybrid or coated systems. There is a literature report in which no PFPE lubricant degradation is observed when one of the bearing components is coated with TiC or ZrO_2 [15]. This is in contrast to our findings, in which lubricant degradation is life limiting, even with all the wear components coated with TiN. However, in our findings, the lubricant degradation mechanism for the coated components is believed to be thermal as opposed to catalytic. The observed differences may be the result of differences in the testing. In our test, the lubricant was exposed to rather severe tribological conditions. Antiwear additives, once they are developed for the PFPE fluids, would be expected to further enhance performance by lowering contact temperatures, resulting in a reduction in the rates of catalytic and thermal degradation. This is a topic for further research.

IV. CONCLUSIONS

The results of our testing lead to the following conclusions.

- 1. The performance of PFPE fluids under boundary lubrication conditions can be extended through the use of ceramic or hard-coated components instead of stainless steel.
- 2. The amount of improvement in wear life appears to be limited by the onset of thermal degradation, which is material independent. In our apparatus, the maximum improvement was approximately a factor of 10.

REFERENCES

- 1. D. J. Carre, Trib. Trans., 31 (1988) 437.
- 2. D. J. Carre, J. Syn. Lubr., 6 (1989) 3.
- 3. D. J. Carre and J. A. Markowitz, ASLE Trans., 28 (1985) 40.
- 4. S. M. Hsu and E. E. Klaus, ASLE Trans., 21 (1978) 201.
- 5. D. J. Carre, ASLE Trans., 29 (1986) 121.
- 6. K. L. Paciorek, D. H. Harris, J. H. Nakahara, M. E. Smythe, and R. H. Kratzer, J. Fluorine Chem., 29 (1985) 399.
- 7. F. B. McKee, Proc. 1987 Symp. on Ball Bearing Tech., Orlando, Florida, 1987.
- 8. H. J. Boving, W. Hanni, H. E. Hintermann, F. Fluhmann, and M. Waelti, Proc. 1987 Symp. on Ball Bearing Tech., Orlando, Florida, 1987.
- 9. R. A. Hanson, Proc. 1987 Symp. on Ball Bearing Tech., Orlando, Florida, 1987.
- 10. H. J. Boving, H. E. Hintermann, and G. Stehle, Lubr. Eng., 37 (1981) 534.
- 11. B. H. Baxter and B. P. Hall, Proc. 19th Aerospace Mech. Symp., NASA Conf. Publ. 2371, 1985, p. 179.
- 12. R. J. Roark and W. C. Young, <u>Formulas for Stress and Strain</u>, McGraw-Hill, New York, 1975, p. 517.
- 13. S. van der Zwaag and J. E. Field, Philosophical Mag., A, 46 (1982) 133.
- 14. E. Torok, A. J. Perry, L. Chollet, and W. D. Sproul, Thin Solid Films, 153 (1987) 37.
- R. A. Rowntree and M. J. Todd, Materials Research Soc. Proc., 140 (1989) 21.

LABORATORY OPERATIONS

The Aerospace Corporation functions as an "architect-engineer" for national security projects, specializing in advanced military space systems. Providing research support, the corporation's Laboratory Operations conducts experimental and theoretical investigations that focus on the application of scientific and technical advances to such systems. Vital to the success of these investigations is the technical staff's wide-ranging expertise and its ability to stay current with new developments. This expertise is enhanced by a research program aimed at dealing with the many problems associated with rapidly evolving space systems. Contributing their capabilities to the research effort are these individual laboratories:

Aerophysics Laboratory: Launch vehicle and reentry fluid mechanics, heat transfer and flight dynamics; chemical and electric propulsion, propellant chemistry, chemical dynamics, environmental chemistry, trace detection; spacecraft structural mechanics, contamination, thermal and structural control; high temperature thermomechanics, gas kinetics and radiation; cw and pulsed chemical and excimer laser development, including chemical kinetics, spectroscopy, optical resonators, beam control, atmospheric propagation, laser effects and countermeasures.

Chemistry and Physics Laboratory: Atmospheric chemical reactions, atmospheric optics, light scattering, state-specific chemical reactions and radiative signatures of missile plumes, sensor out-of-field-of-view rejection, applied laser spectroscopy, laser chemistry, laser optoelectronics, solar cell physics, battery electr chemistry, space vacuum and radiation effects on materials, lubrication and surface phenomena, thermionic emission, photosensitive materials and detectors, atomic frequency standards, and environmental chemistry.

Electronics Research Laboratory: Microelectronics, solid-state device physics, compound semiconductors, radiation hardening; electro-optics, quantum electronics, solid-state lasers, optical propagation and communications; microwave semiconductor devices, microwave/millimeter wave measurements, diagnostics and radiometry, microwave/millimeter wave thermionic devices; atomic time and frequency standards; antennas, rf systems, electromagnetic propagation phenomena, space communication systems.

Materials Sciences Laboratory: Development of new materials: metals, alloys, ceramics, polymers and their composites, and new forms of carbon; nondestructive evaluation, component failure analysis and reliability; fracture mechanics and stress corrosion; analysis and evaluation of materials at cryogenic and elevated temperatures as well as in space and enemy-induced environments.

Space Sciences Laboratory: Magnetospheric, auroral and cosmic ray physics, wave-particle interactions, magnetospheric plasma waves; atmospheric and ionospheric physics, density and composition of the upper atmosphere, remote sensing using atmospheric radiation; solar physics, infrared astronomy, infrared signature analysis; effects of solar activity, magnetic storms and nuclear explosions on the earth's atmosphere, ionosphere and magnetosphere; effects of electromagnetic and particulate radiations on space systems; space instrumentation.